

Abstract

The goal of the AIAA DBF 2019-2020 team is to build and iterate an airplane that annihilates the opposition in the AIAA DBF competition in April of 2020. The airplane would have scored as many points as possible by effectively and efficiently completing all missions laid out by the DBF 2020 Rules. The objective for the 2020 DBF competition is to design, build, and test a banner-towing, charter aircraft. To build the most effective plane, subcomponent concepts for the plane must be chosen. These subcomponents include fuselage, wings, restraint system, landing gear, propeller, motor, and battery. After the components are chosen, detailed analysis of each component will be completed to ensure that the optimal design has been chosen for the plane. The overall budget for the aircraft is \$3,200.

Design Solution

Design was based on plane sizing guide from M. H. Sadraey's Aircraft Design Textbook.



Test 1:

- Wing Tip Test
- Simulates Competition Structural Test
- Plane is held by both end of wing
- Plane must not fracture to complete wing tip test

Test 2:

- First Flight Test
- Gathered data for takeoff distance
- Takeoff distance was approximately 4.9 meters, satisfying the takeoff requirement.



AIAA Design, Build, Fly 2019-2020 Mechanical and Aerospace Engineering Brenton Cherry, Trevor Haessig, Shauryya Mishra, Jared Thomas, Vilis Vuskalns, Dr. Koch **Customer Needs and Requirements Concept Selection** • The aircraft must take off in 6.1 meters (Missions 1 & 3) The aircraft must carry as many passengers as possible (Mission 2) The aircraft must deploy, tow (up to 10 min), and release a banner with minimum length of 25 cm while in flight (Mission 3) Maximum allowable wingspan is 152 cm The aircraft must fly the path below for the competition The aircraft must complete 3 laps in 5 minutes (Missions 1 & 2) 360 Degree Turn Starting Line meters meters Manufacturing Fuselage Wing Tail • Foam Leading Edge Plywood cross-Carbon Fiber Lfor applying sections lined up brackets to connect Monokote Balsa sheets glued to tail to boom Ailerons to control cross sections Balsa control surfaces roll connected with • Balsa ribs with basswood spars hinges

Testing and Validation

Test 3:

- Second Flight Test
- Tried to test plane maneuverability
- Due to elevator issue, plane suffered a crash shown below

Test 4:

- Final Flight Test
- Tested plane endurance and speed
- Plane flew for approximately 3 minutes at 55 ft/s before batteries needed charged





- for landing gear
- the tail design
- system
- mechanism







Final Plane Design

Acknowledgments

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Tricycle configuration chosen Conventional tail chosen for Velcro strap chosen for passenger/luggage restraint Claw mechanism chosen for banner deployment At3520 Motor chosen for propulsion system